## General Guidelines for Crash Analysis in LS-DYNA

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### • Element Shapes

- Avoid use of triangular shells, tetrahedrons, pentahedrons whenever possible (ok in rigid bodies).
- ESORT=1 if triangular shells are present,\*CONTROL\_SHELL.
- ESORT=1 if tetrahedrons, pentahedrons are present,\*CONTROL\_SOLID.

## Warping Stiffness in Shells

• Warped shells are too soft.



- For warped B-T shells, set BWC=1 and invoke the more costly full projection for warping stiffness (PROJ=1) since drill projection inhibits rigid body rotation.
- For fully-integrated shell ELFORM=16, set hourglass formulation IHG to 8 to invoke warping stiffness.



## • Shells

- Invoke invarient node numbering (\*CONTROL\_ACCURACY) so that results are insensitive to the order of the nodes in the element connectivity.
- Shell thickness update (ISTUPD in \*CONTROL\_SHELL) is generally not need for crash analysis (req'd for metal forming).
- Use minimum of 3 integration points through the thickness (NIP) for shell parts undergoing plastic deformation.
- Set shear factor in \*SECTION\_SHELL to theoretical value of 5/6.
- Turn on bulk viscosity for shells via \*CONTROL\_BULK\_VISCOSITY

## Solids

 Use ELFORM=1 for solids and include appropriate hourglass control



#### **Invariant Node Numbering**





invariant node numbering invoked



## Hourglass Control

- If using \*CONTROL\_HOURGLASS, define part-specific \*HOURGLASS cards to overwrite the global hourglass definitions where appropriate.
  - Be careful when importing validated barrier/head models
- Recommend stiffness hourglass control, IHQ=4, with hourglass coefficient QM = 0.03 for metal and plastic parts.
- Recommend viscosity-based hourglass control for foams and rubbers (IHG=2 or 3) or hourglass formulation 6
  - In soft materials, stiffness-based hourglass control (IHG=4 or 5) causes overly stiff response even with a reduced hourglass coefficient.



## • Materials

- When including strain rate effects in plasticity models, set VP=1.
  - Uses plastic strain rate rather than total strain rate.
  - Results in smoother response
- Stress-strain curves should be smooth, especially for foams.
- Mass of null shells and null beams is included in total mass.
  - Unless additional mass is intentional, set density of null shells and beams to a small value.
- Curves defining constitutive data should have abscissa values in the anticipated working range. Curves will be extrapolated by LS-DYNA if necessary.



## Connections

- Nodal rigid bodies
  - Avoid 1-noded RB's and nodal rigid bodies with numerically insignificant inertia as these rigid bodies are deleted and a warning is issued to the D3HSP file.
- Joints
  - Joint node pairs should be a reasonable distance apart.
  - When increasing joint penalty factor to take out 'slop' in penalty-based joint, the time step scale factor may need to be reduced to avoid instability.



#### Connections

- Discrete springs
  - Spring nodes cannot be massless.
  - If NON\_LINEAR spring material is used, define stiffness in compression and tension.
  - Use only N1 to N2 orientation.
- Deformable spotwelds
  - Avoid "free/suspended" spotwelds.
  - Look out for spotweld nodes that are not tied (see warnings in d3hsp).
  - Exclude spotwelds from contact (automatic if MAT\_SPOTWELD is used)
  - Invoke stiffness damping in shells if using \*CONTACT\_SPOTWELD\_TORSION
  - Solid spotwelds show promise.
    - Pro: Less sensitive to spotweld placement
    - Con: No automatic spotweld generation



### • Rigid Bodies

- Refined mesh of rigid bodies encouraged.
  - Added expense is minimal.
  - More realistic mass properties and distribution of contact forces.
- Specify reasonable elastic constants for \*MAT\_RIGID, e.g., those of steel.
  - Affects contact stiffness unless SOFT=2.
- Do not impose constraints on nodes of rigid bodies. Impose constraints on card 2 of \*MAT\_RIGID.



## **Inertia Considerations: Example**

Cubes with Applied Moments Time = 0



shells solids



# MMI: Mass Moment of Inertia units: mm, kg, ms

Number of elements		Element	LS-DYNA calculated		Actual MMI	% Error
shells	solids	iengui	shells	solids	(solid)	Solids
6	1	25	38.38	38.38	12.79	200%
24	8	12.5	25.59	19.19	12.79	50%
96	64	6.25	22.39	14.39	12.79	12.5%
384	512	3.125	21.59	13.19	12.79	3.1%
1536	4096	1.5625	21.39	12.89	12.79	0.8%



## Initial Velocity

- Be careful with rigid body initial velocities.
  - \*PART\_INERTIA supercedes all other initial velocity commands.
  - If initial velocity of rigid bodies is inexplicably off, use double precision or use \*INITIAL\_VELOCITY\_RIGID command.
- Make final check of initial velocity with a plot of velocity vectors at time = 0.

# **Modeling Guidelines for Crash Analysis**

## Contact

- Take care to account for shell thickness when generating the mesh.
- Avoid redundant contact definitions.
- Use only AUTOMATIC contacts.
- Premature nodal release from contact may lead to inconsistent answers. Increase contact thickness for very thin shells
- Use of IGNORE=1 is encouraged for contact in cases where small initial penetrations are reported.
- Use of SOFT=1 is preferred over SOFT=0, especially in treating contact between dissimilar materials.
- Use SOFT=2 for contact surfaces with sharp corners.
- Use AUTOMATIC\_GENERAL for beam-to-beam contact.



## Segment-Based Contact (SOFT=2)





#### Postprocessing

- Animate results to check for nonphysical behavior, for example, parts noticeably penetrating other parts.
- Check energies in **GLSTAT** and **MATSUM** 
  - Use \*CONTROL\_ENERGY to turn on computation of relevant energy values.
  - Energy ratio should remain close to 1.0.
  - Hourglass energy < 10% of peak internal energy.
  - If no contact friction, contact energy in GLSTAT should be relatively small.
  - If contact friction is nonzero, contact energy should be positive and not necessarily small.
- System added mass should be < 1% of physical mass (check GLSTAT)



Automotive Crash guidelines are a good start.

## Differences in Automotive Crash and Aerospace Impact?

- Impact velocities and thus strain rates are higher in Aerospace applications.
  - Material strain rate effects
  - Material damage/failure/erosion
- Aerospace materials are generally lighter, stiffer, more complex.

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## Additional guidelines for Aerospace impact

- Time Step
  - Reduce time step scale factor to 0.6 or 0.7.
- Hourglass Control
  - Look at viscous-based hourglass control as first alternative.
  - Use type 1 hourglass control with hourglass coefficient=1.E-3 for fluids.
- Bulk Viscosity
  - Increase bulk viscosity coefficients by a factor of 10 for bird material.



## Additional guidelines for Aerospace impact

- Contact
  - When elements are eroded, use \*CONTACT\_ERODING with SOFT=1.
    - Eroding\_nodes\_to\_surface for Lagrangian bird strike.
    - Eroding \_single\_surface otherwise
  - Retain mass of eroded elements via ENMASS in \*CONTROL\_CONTACT
- Materials
  - Include strain rate effects if data is available.
  - For sandwich composites, invoke laminate shell theory (LAMSHT in \*CONTROL\_SHELL).
- ...



## **Effect of Deleted Nodes on Contact**

